

# Transition modelling in the CIRA UZEN flow solver

1<sup>st</sup> AIAA Transition Modeling and Prediction Workshop

D. de Rosa, CIRA (Italian Aerospace Research Centre) d.derosa@cira.it



#### **Outline**

- > Flow solver description
- ➤ Case 0 Fully Turbulent 3D Bump in Channel
- ➤ Case 1 Zero-Pressure-Gradient Flat Plate
  - Case 1A Grid resolution on T3A
  - Case 1B Grid resolution on T3B
- $\triangleright$  Case 2 NLF(1)-0416 Airfoil
  - Case 2A Grid resolution at  $\alpha = 0^{\circ}$ ,  $5^{\circ}$
  - Case 2B Polar on medium grid
  - Case 2C Polar on fine grid



#### **RANS Flow solver**

#### **UZEN Code**

- Flow Solver for steady and unsteady Euler, and RANS equations
- > Spatial Discretization
  - Structured Multi-Block, Finite Volume
  - Cell Centered with blended 2nd and 4th order artificial dissipation
- ➤ Dual-Time Stepping for unsteady flows
- ➤ Time Advancement for steady flows
  - Runge-Kutta with multigrid, local time-stepping, residual averaging
- ➤ Turbulence Models
  - Baldwin-Lomax
  - Spalart-Allmaras
  - Myong-Kasagi, NLEV (Shih formulation) κ-ε
  - κ-ω : Wilcox, Kok TNT, Menter BSL and SST, SST-LR
  - DES for SA and κ-ω SST
  - XLES for κ-ω TNT



#### **Transition Model**

Transport equation for intermittency

$$\frac{\partial \rho \gamma}{\partial t} + \frac{\partial \rho U_{j} \gamma}{\partial x_{j}} = P_{\gamma} - D_{\gamma} + \frac{\partial}{\partial x_{j}} \left[ \left( \mu + \frac{\mu_{t}}{\sigma_{\gamma}} \right) \frac{\partial \gamma}{\partial x_{j}} \right]$$

$$P_{\gamma} = F_{length} \rho S \gamma (1 - \gamma) F_{onset}$$

$$D_{\gamma} = c_{a2} \rho \Omega \gamma F_{turb} (C_{e2} \gamma - 1)$$

Triggering functions (analytical)

SST turbulence model interaction ( $P_k^{lim}$  additional production term)

$$\frac{\partial \rho \kappa}{\partial t} + \frac{\partial \rho U_j \kappa}{\partial x_j} = \gamma P_{\kappa} + P_{\kappa}^{lim} - \max(\gamma, 0.1) D_{\kappa} + \frac{\partial}{\partial x_j} \left[ (\mu + \sigma_{\kappa} \mu_t) \frac{\partial \kappa}{\partial x_j} \right]$$

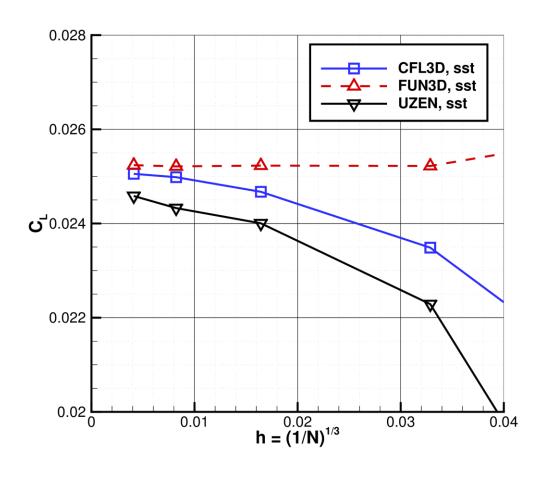
- ➤ No sustaining terms
- ➤ Limiter on dissipation disabled

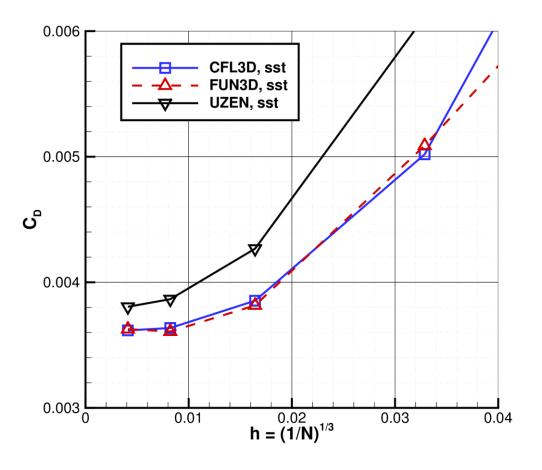
Menter, F.R., Smirnov, P.E., Liu, T. *et al.* A One-Equation Local Correlation-Based Transition Model. *Flow Turbulence Combust* **95**, 583–619 (2015). https://doi.org/10.1007/s10494-015-9622-4



#### Case 0: Fully Turbulent 3D Bump-in-Channel

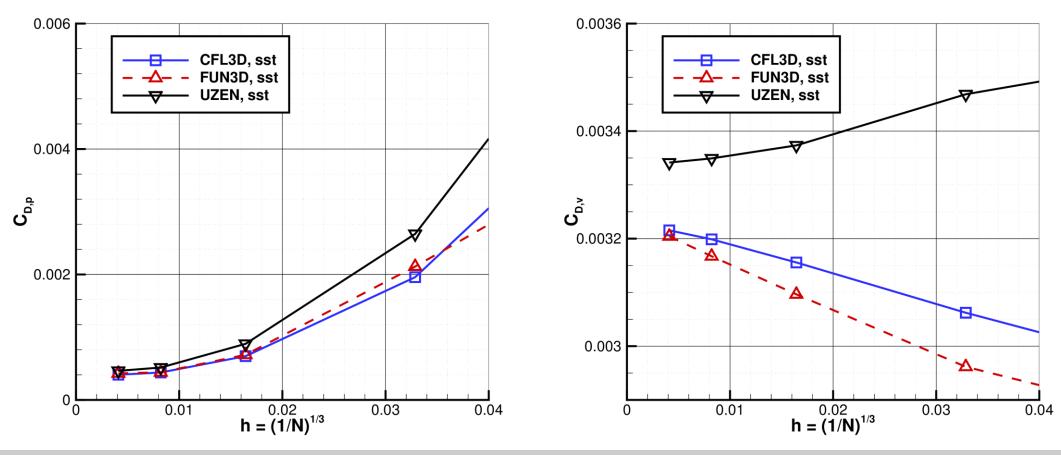
Grids: PLOT3D structured version







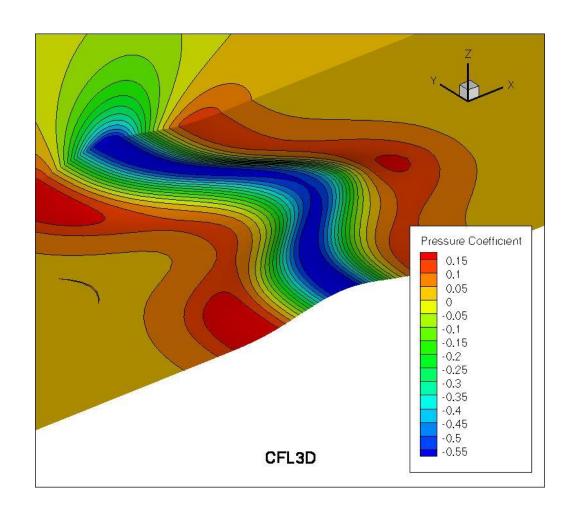
### Case 0: Fully Turbulent 3D Bump-in-Channel

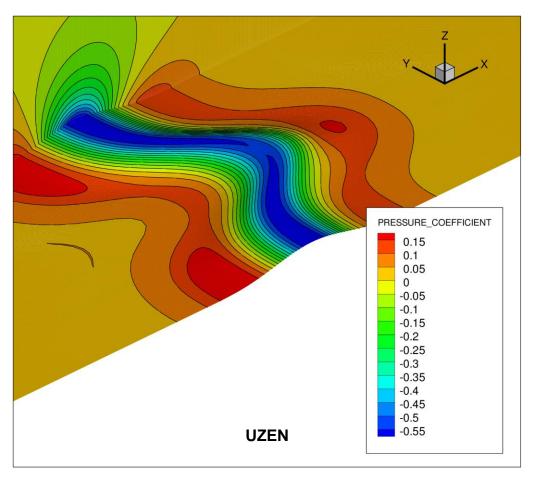


Code	Quantity	Computed apparent order, p	Approx rel fine-grid error, e <sub>a</sub> <sup>21</sup>	Extrap rel fine-grid error, e <sub>ext</sub> <sup>21</sup>	Fine-grid convergence index, $GCI_{fine}^{21}$
CFL3D	$C_d$	3.49	0.532%	0.052%	0.065%
UZEN		2.3	1.93%	0.283%	0.353%



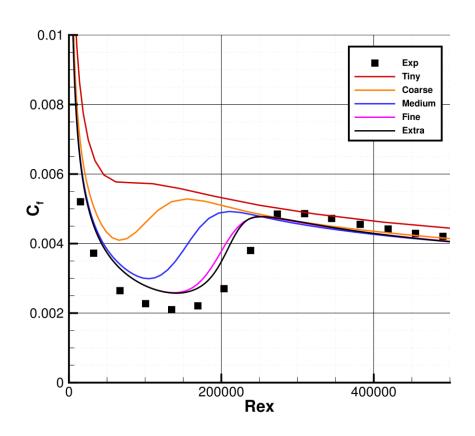
### Case 0: Fully Turbulent 3D Bump-in-Channel

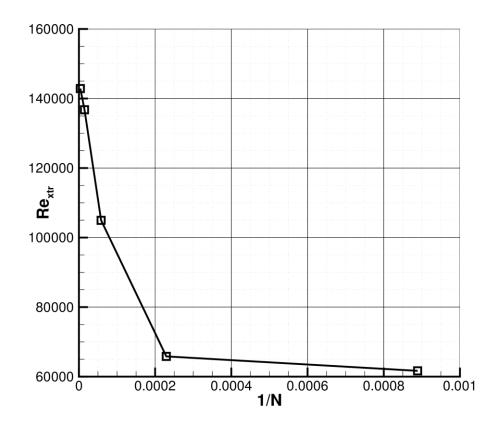






#### Case 1A: T3A ZPG

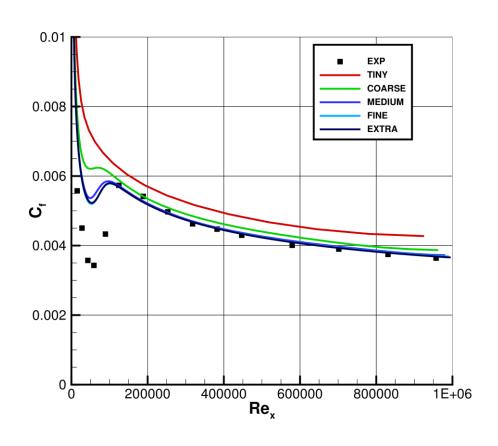


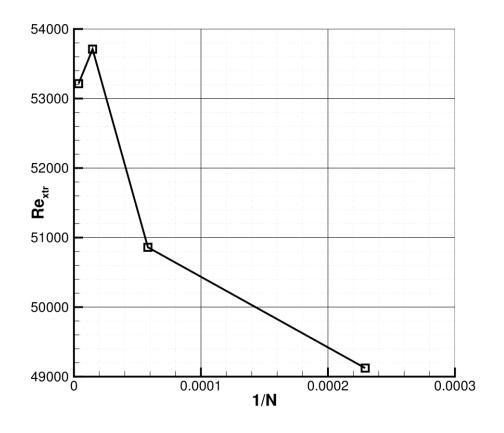


- ightharpoonup Improved  $x_{tr} \sim \min(c_f)$  as grid refines
- $\triangleright$  Laminar  $c_f$  higher w.r.t experimental data



#### Case 1B: T3B ZPG

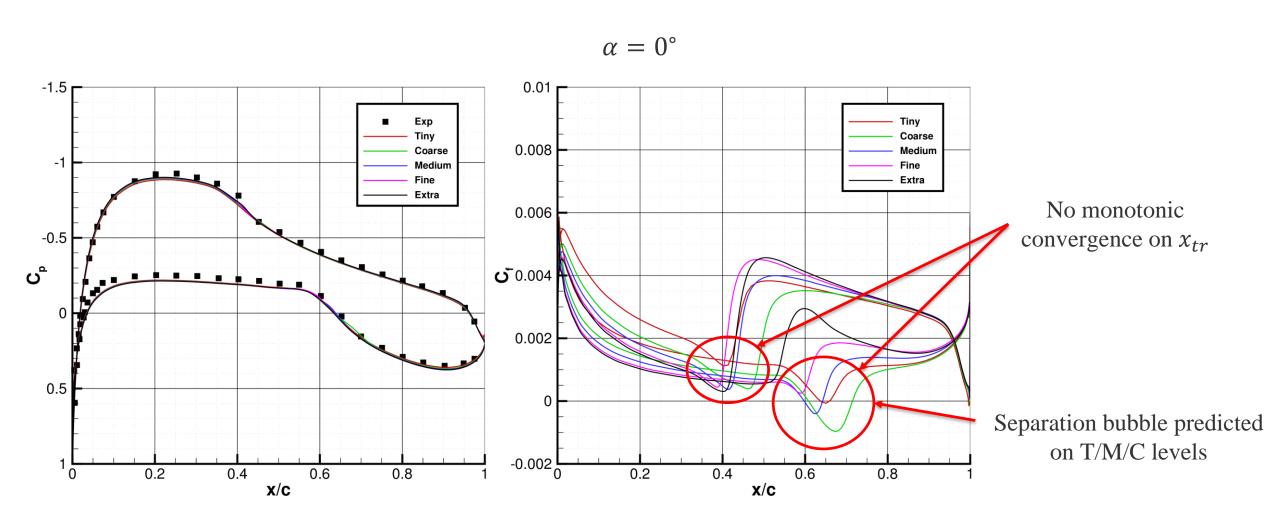




- ightharpoonup Improved  $x_{tr} \sim \min(c_f)$  as grid refines
- $\triangleright$  Laminar  $c_f$  higher w.r.t experimental data

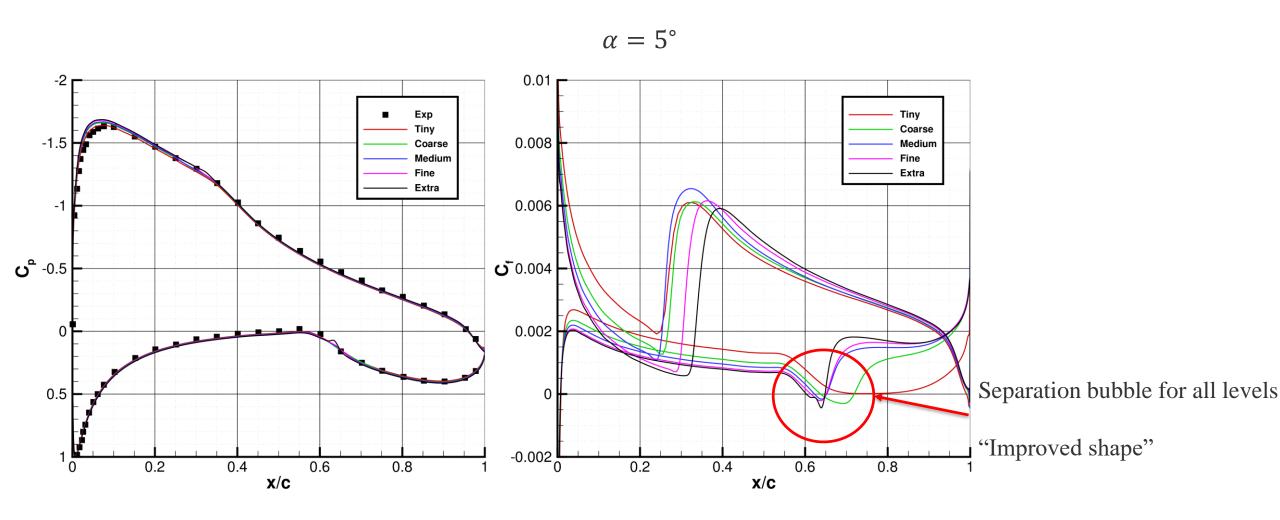


#### Case 2A: NLF(1)-0416 Airfoil



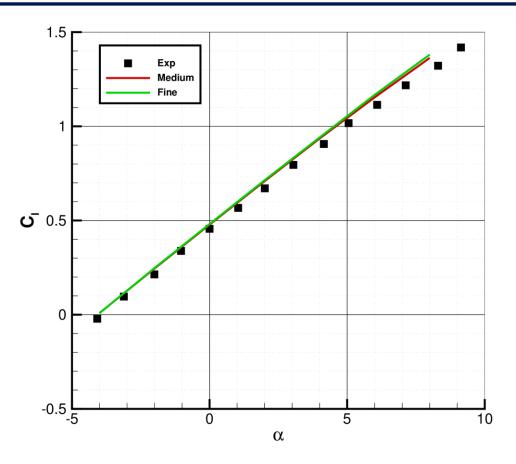


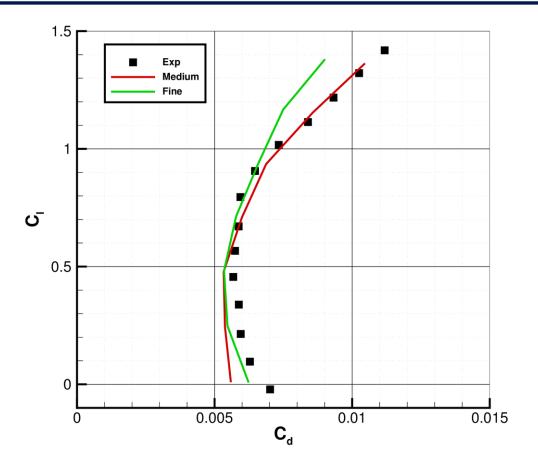
## Case 2A: NLF(1)-0416 Airfoil





#### Case 2B/C: NLF(1)-0416 Airfoil

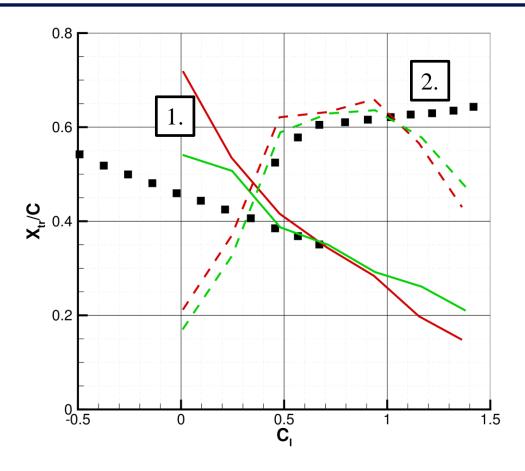


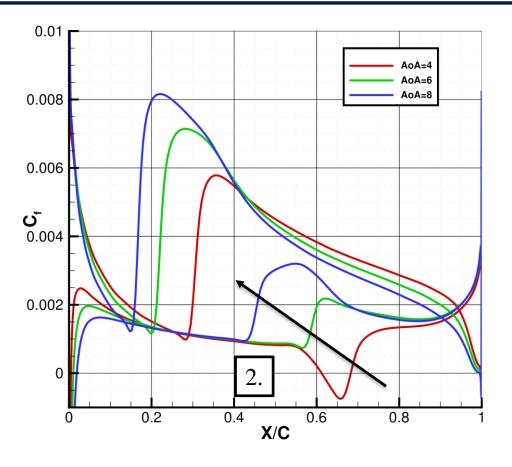


- ➤ Good agreement w.r.t experimental data
- $\triangleright$  Solutions on fine grid underestimate  $C_d$  at high  $\alpha$



#### Case 2B/C: NLF(1)-0416 Airfoil





- 1. Overestimation on the upper side for low  $\alpha < 0^{\circ}$
- 2. Misprediction on the lower side for  $\alpha \ge 6^{\circ}$



#### **Conclusions**

#### **ZPG** Flat Plate

- Improvement in solution refining the grid
- High  $c_f$  values w.r.t. experimental data in the laminar region (especially in T3B)

#### **NLF Airfoil**

- Grid convergence is not clearly achieved
  - $\triangleright$  Refining the mesh lowers  $c_f$  values in laminar region (expected)
  - $\triangleright$  No-monotonic shift for the transition point (assumed as min $(c_f)$ ).
    - ☐ Due to the threshold function?
- Transition point is predicted downstream w.r.t experimental data
  - $\triangleright$  Misprediction on the lower side for  $\alpha \ge 6^{\circ}$



# Thanks for your attention!